

MODELING STRESS INTENSITY FACTORS OF CRACKS EMENATING FROM A CENTERED CIRCULAR HOLE IN A TUNGSTAN-CARBIDE RECTANGULAR PLATE UNDER UNIAXIAL COMPRESSION

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Abstract

Crack propagation under load is different than displacement controlled. The displacement controlled loads usually cause stable propagation. This paper evaluates the mode I Stress Intensity Factors (SIF) of a crack emanating from a centered circular hole in a square plate under compression. Such a geometry is similar to square cutting tool insert. The plate is made from tungsten carbide (WC+Co). Two types of compression loads are considered: pressure traction and normal displacement. Linear 3D static analyses are conducted with the aid of a commercial software (Abaqus/CAE 2020). The SIFs for various crack lengths are obtained using the J integral and verified by analytic expansion of the crack's normal gaps. The results show that displacement load causes lower SIFs and a more stable propagation. For some loads the crack might come to a halt even for traction load.

1. Introduction

A brittle material such as tungsten carbide have no yield strength, and involves fracture without any appreciable plastic deformation. We are interested in understanding the propagation of a crack emanating from a centered circular hole in a square plate under compression as shown in Fig. 1a. This geometry simulates a square insert cutting tool as shown in Fig. 1b.

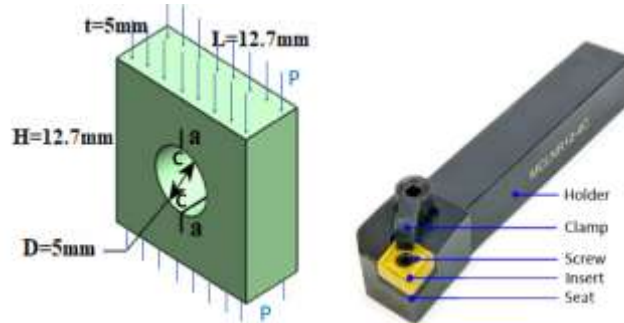


Fig. 1: a. A crack (a) emanating from a circular hole (D) of a rectangular tungsten-carbide plate (H,L,t) under compression P. b. Tool holders for a Common square single cutting tool insert

Seven crack lengths were modeled numerically ($a = 0.5, 1.0, 1.5, 2.0, 2.5, 3.0, 3.5$ mm) under pressure ($P = 705$ MPa) and normal displacement ($d = -25.689$ μ m). These compression loads create the same circumferential tension stress for $a = 0$ at point C (figure 2b) at the middle of the plate ($t_m = 2.5$ mm) where plane strain conditions prevail. Linear 3D static analyses were conducted using Abaqus/CAE 2020.

2. Results

Due to symmetry only 1/8 of the geometry shown in figure 1a was modeled numerically. The numerical mesh used quadratic hexahedral elements of type C3D20. A typical mesh is shown in Fig. 2a for $a = 0.5$ mm. The mesh was made locally denser near the crack tips having seed size of 62.5 μ m. All meshes were checked for numerical convergence.

The two cases of loading ($P = 705$ MPa and $d = -25.689$ μ m) were first applied for $a = 0$ (no crack). The tensional circumferential stress σ_{xx} between points C and E is shown in Figure 2b. It can be observed that the stress at point C due to both load cases is equal: $\sigma_{xx}^* = 1.807$ GPa. The fact that both normal stresses

variations become negative (compression) at ~ 2.3 mm indicate that for crack lengths greater than 2.3mm, the cracks might become stable (will not advance). The insert within Fig. 2b show the contour map of σ_{xx} indicating that the stress is higher within the plate than on the free face.

A Comparison of the normalized K_I^* versus normalized crack length a^* is shown in Fig. 2c. The mode I SIFs were calculated using J integral. The SIFs due to pressure load were verified by using expansion of the crack face normal gaps using 3 terms of the analytic asymptotic expansion. The maximum relative difference was less than 3.7% for $0.5 \text{ mm} \leq a \leq 3 \text{ mm}$ which confers reliability to the results. The SIF was normalized by $K_I^* = K_I / K_I^0$ where $K_I^0 = 1.12\sigma_0\sqrt{\pi a_0}$ which is the solution of an edge crack in semi-infinite region subjected to tension. A short crack $a_0 = 50 \mu\text{m}$ is assumed and $\sigma_0 \equiv \sigma_{xx}^* = 1.807 \text{ GPa}$. The normalized crack length is $a^* = 2a / (H - D)$. It can be observed that $\partial K_I^* / \partial a^* < 0$ for application of constant displacement and becomes negative for constant pressure at crack length $a^* = 0.4$.

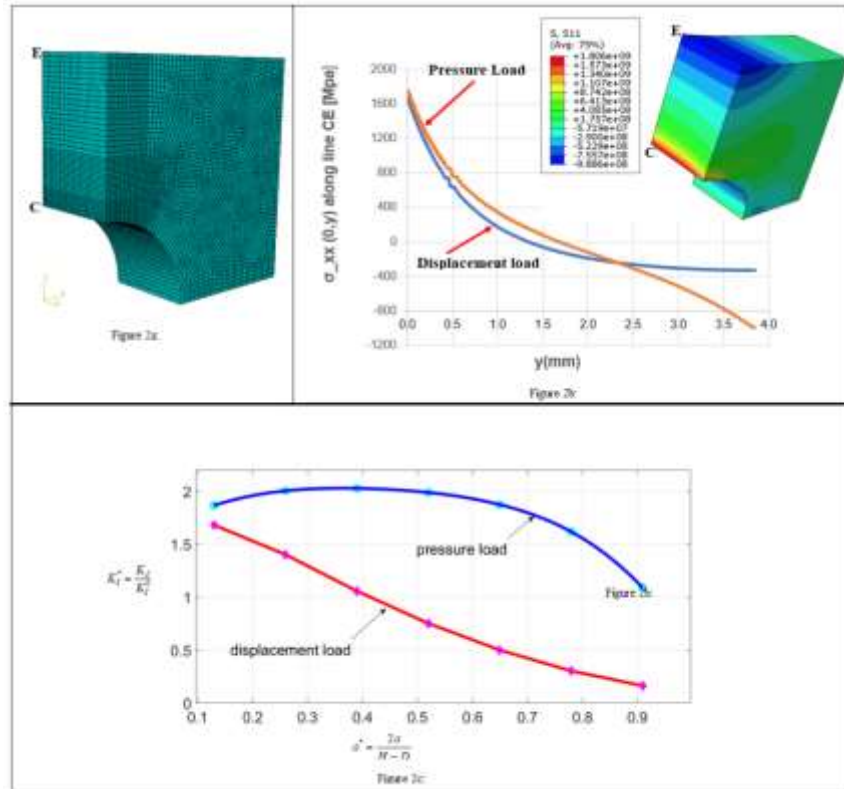


Fig. 2: a. A typical mesh for $a = 0.5$ mm. b. Comparison of normal stress along the line CE for constant pressure and constant displacement load without cracks. c. Comparison of normalized SIF (K_I^*) versus normalized crack lengths (a^*) due to constant pressure load and constant displacement load.

3. Conclusions

- Constant displacement is preferable type of load because $\partial K_I^* / \partial a^* < 0$ for all a^* which lead to stability.
- Even a crack loaded by pressure control might become stable for long cracks because $\partial K_I^* / \partial a^* < 0$ for $a^* \geq 0.4$.

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